

UNITED STATES NAVAL ACADEMY
Annapolis, Maryland 21402-5042
DIVISION OF ENGINEERING AND WEAPONS

Report EW-05-00

COMMUNICATION, FEEDBACK, AND
DECENTRALIZED CONTROL FOR
PLATOONS OF UNDERWATER VEHICLES

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11 August 2000

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1 Introduction

Many researchers have conjectured that platoons of cooperating mobile robots or autonomous vehicles provide significant benefits over single-unit approaches for a variety of tasks. Further, cooperating robots or vehicles need not necessarily be sophisticated or expensive to out-perform many advanced independent units for tasks such as material transport, scouting, etc. Unfortunately, large-scale control systems for platoons of cooperating mobile robots or autonomous vehicles are difficult to design for real-world situations. Communication requirements, especially with regard to bandwidth limits, are often challenging obstacles to control system design. In this work, we are interested in the development of a design methodology and analysis technique for controlling platoons of autonomous vehicles with a focus on understanding communication requirements for such systems. We present a decentralized control framework applicable to platoons of mobile robots or vehicles and, for illustration, consider a simplified design example for a platoon of autonomous underwater vehicles (AUVs). An underwater example is chosen to highlight the need for control strategies that address reduced communications since communication bandwidth is severely limited underwater.

There are two primary schools of thought on methods for controlling platoons of cooperating mobile robots and vehicles: the system-theoretic and behavior-based approaches. Behavior-based methods (sometimes referred to as reactive control) rely on the use of algorithmic behavior structures without an explicit mathematical model of the subsystems or the environment (see, e.g., [1], [2]). The system-theoretic approach, on the other hand, relies strongly on the use of system dynamics and models of the interactions between the vehicles themselves, as well as between the vehicles and the environment (e.g., [3], [4]).

These two approaches to cooperation have fundamentally differing benefits; neither presents a universal solution to the problem of designing cooperating platoons of autonomous vehicles. The benefits of the system-theoretic approach are that the results are provable and predictable and there are analytic solutions to questions regarding performance. The drawback to system-theoretic techniques is that they are encumbered by the need to approximate

complex dynamics, models for which are never fully accurate. A benefit of behavior-based approaches is that they are motivated by biological systems that have shown great survivability and adaptability and can exhibit significant, if hard to quantify, performance and robustness. Behavior-based approaches do not, however, readily admit a closed-form design process and are sometimes as likely to exhibit unexpected and undesirable behavior as they are to perform as desired. Herein, we will focus on systems-theory-based approaches, as we wish to develop rigorous design methodologies while addressing issues of limited communications bandwidth.

Fundamental to the problem of cooperation is the question of communication: How, what, and when should robots communicate to achieve a given task? In our work, we make a distinction between *implicit* information, such as reaction forces experienced by two robots cooperatively handling a rigid structure, and *explicit* information for which dedicated communications bandwidth must be utilized. The effect of communication on cooperative behaviors in mobile systems has been studied extensively in the framework of reactive robot architectures [5].

Control-theoretic methods for cooperating platoons of robots in use today rely on decentralized control almost exclusively, although not in the formal framework developed in [6] (see, e.g., [3], [4]). These systems typically use local measurements and implicit communications as feedback for local controllers. Herein we propose that the formal framework of decentralized control may offer a more rigorous design procedure for these types of systems and, in addition, offer a strategy for determining what types of explicit communication are required.

A strong component of our results is that the design technique and analysis scale to arbitrarily large platoons of cooperating vehicles. Indeed, the communication that is required among vehicles in the platoon is independent of the number of vehicles in the platoon.

Control objectives for cooperating platoons of robots have typically consisted of generating specific formations [7] or global behaviors [3] based on local relationships (e.g., the exact location of the nearest neighbor). These types of objectives have typically required

high bandwidth communications under the informal decentralized control schemes used. We choose to control global functions of a platoon, such as the center (average position) of the platoon and the distribution of vehicles about the center. Vehicles are not commanded to be in specific positions. Instead, the vehicles autonomously move to locations that satisfy the center and distribution commands under a decentralized control law with surprisingly little inter-vehicle communication. This removes the requirements, commonly seen in formation control problems, that each vehicle observe, measure, or receive (via explicit communication) the state of the entire system. Further, we propose that this framework allows a large amount of flexibility that is difficult to encode in traditional formation frameworks.

2 Background and Problem Statement

We consider a platoon composed of r heterogeneous vehicle subsystems, each described by the dynamics

$$\dot{x}_i(t) = f_i(x_i(t), u_i(t)) \quad (1)$$

where $x_i \in \mathbb{R}^{n_i}$, $u_i \in \mathbb{R}^{m_i}$, and $i = 1, \dots, r$. The dynamics of the platoon are completely uncoupled and all interaction between subsystems must be in the form of either implicit or explicit communication. Each subsystem has a local controller that generates the local control signal $u_i(t)$ based on measured signals produced by the subsystem or on signals communicated to the subsystem from elsewhere. This is a *decentralized* control structure, and we borrow heavily from the decentralized control literature in the analysis and design of the platoon controller. Early work on the existence of decentralized controllers, as in [6], [8], and [9], develops essential tools that we use in determining what communication is required between subsystems for a decentralized controller to exist. Since our primary interest is in examining communications structures, we take a local viewpoint and design linear decentralized controllers. Through simulation of a nonlinear platoon model, we find that the linear controllers perform well.

2.1 Regulation of Platoon-Level Functions

Our objective is to regulate *platoon-level* functions, such as the average position of the vehicles in a platoon or the distribution of the vehicles about the average position. The platoon-level function is denoted $h_c(x_1, \dots, x_r) \in \mathbb{R}^{p_d}$, a function of the entire platoon state.

We adopt the working assumption that only a single vehicle has the capability to measure the platoon-level function or, equivalently, that the platoon-level function is measured by an exogenous system and transmitted to one vehicle in the platoon. Certainly there are practical considerations involved with this assumption, including issues related to single-point failure. Yet this assumption allows for the use of active sensors for the measurement since no crosstalk is present to degrade performance. Within our framework, different vehicles are permitted to have different dynamics and sensor suites and this certainly includes a single unit possessing a single copy of a potentially expensive component. For platoon-level measurements such as average vehicle position and distribution, fine-grained measurements (i.e., exact position of each vehicle) are unnecessary—only measurements that indicate vehicle position density are required.

Without loss of generality, the platoon-level functions are assumed to be measured by subsystem 1. In other words, we assume that $h_c(x_1, \dots, x_r)$ is an output of subsystem 1. To ensure zero steady-state tracking error, integrators are connected in series to the output representing the global functions, yielding a new state variable

$$\dot{q}(t) = h_c(x_1(t), \dots, x_r(t))$$

where $q(t) \in \mathbb{R}^{p_d}$ is the integrator state.

2.2 Platoon

Again, the platoon is the parallel connection of the r subsystems. Since our control system is designed from a local decentralized viewpoint, we linearize the platoon dynamics (1) at

an equilibrium value of the subsystem states and inputs and write, somewhat loosely,

$$\dot{\bar{x}}(t) = \begin{bmatrix} 0 & H_c \\ 0 & F \end{bmatrix} \bar{x}(t) + \begin{bmatrix} \hat{G}_1 & \dots & \hat{G}_r \end{bmatrix} \bar{u}(t) \quad (2)$$

where

$$F = \begin{bmatrix} F_1 & 0 & \dots & 0 \\ 0 & F_2 & & \\ \vdots & & \ddots & \\ 0 & & & F_r \end{bmatrix} \in \mathbb{R}^{n \times n}, \quad \hat{G}_i = \begin{bmatrix} 0 \\ \vdots \\ 0 \\ G_i \\ 0 \\ \vdots \\ 0 \end{bmatrix} \in \mathbb{R}^{(n+p_d) \times m_i}, \quad H_c = \begin{bmatrix} H_{c1} & \dots & H_{cr} \end{bmatrix} \in \mathbb{R}^{p_d \times n}$$

and

$$F_i = \frac{\partial}{\partial x_i} f_i(x_i^o, u_i^o) \in \mathbb{R}^{n_i \times n_i}, \quad G_i = \frac{\partial}{\partial u_i} f_i(x_i^o, u_i^o) \in \mathbb{R}^{n_i \times m_i}, \quad H_{ci} = \frac{\partial}{\partial x_i} h_c(x_1^o, \dots, x_r^o) \in \mathbb{R}^{p_d \times n_i}.$$

The partial derivatives are evaluated at equilibrium values of the state and input, x_i^o and u_i^o , and the deviation variables are defined

$$\bar{x}(t) = \begin{bmatrix} q(t) - q^o \\ x_1(t) - x_1^o \\ \vdots \\ x_r(t) - x_r^o \end{bmatrix} \in \mathbb{R}^{p_d+n} \quad \text{and} \quad \bar{u}(t) = \begin{bmatrix} u_1(t) - u_1^o \\ \vdots \\ u_r(t) - u_r^o \end{bmatrix} \in \mathbb{R}^m.$$

The matrix \hat{G}_i is partitioned into $r+1$ blocks, with dimensions $p_d \times m_i$, $n_1 \times m_i$, \dots , $n_r \times m_i$. The first $p_d \times m_i$ submatrix is always zero. The remaining partitioned blocks are all zeros except for entries corresponding to G_i (where G_1 is the second block, etc.). We also define $n = n_1 + \dots + n_r$ and $m = m_1 + \dots + m_r$.

2.3 Output Functions

In the decentralized control framework, a separate controller is designed for each subsystem. The signal measured by the controller for subsystem i is denoted $y_i(t)$ and partitioned to distinguish those components that are generated locally by subsystem i from those that are exogenous, such as implicit or explicit communication from another subsystem. The local signal might be the subsystem's measured position and velocity, whereas the exogenous signal might be the position of another subsystem that must be explicitly communicated. The partitions of $y_i(t)$ are denoted

$$y_i(t) = h_i(x(t)) = \begin{bmatrix} h_{ia}(x_i(t)) \\ h_{ib}(x(t)) \end{bmatrix} \in \mathbb{R}^{p_i}$$

where $h_{ia}(x_i) \in \mathbb{R}^{p_{ia}}$ is locally generated and $h_{ib}(x) \in \mathbb{R}^{p_{bi}}$ is exogenous to subsystem i . For local analysis, we use the notation

$$H_i = \frac{\partial h_i(x^o)}{\partial x} \in \mathbb{R}^{p_i \times n}, \quad H_{ia} = \frac{\partial h_{ia}(x_i^o)}{\partial x_i} \in \mathbb{R}^{p_{ia} \times n_i}, \quad \text{and} \quad H_{ib} = \frac{\partial h_{ib}(x^o)}{\partial x} \in \mathbb{R}^{p_{ib} \times n_i}$$

for $i = 1, \dots, r$.

3 Decentralized Control

In the decentralized control structure, each subsystem is regulated by a separate dynamic output feedback controller

$$\begin{aligned} \dot{z}_i(t) &= A_i z_i(t) + B_i y_i(t) \\ u_i(t) &= C_i z_i(t) \end{aligned} \quad i = 1, \dots, r. \quad (3)$$

where $A_i \in \mathbb{R}^{n_{ki} \times n_{ki}}$ and n_{ki} is the dimension of the controller state. The output signal $y_i(t)$ of subsystem i is the signal measured by the controller. The controller output $u_i(t)$ is the forcing function for subsystem i . There are r such controllers in the platoon—one for each subsystem. Note that there is no interaction between the controllers. All interaction is due

to the subsystem output signal $y_i(t)$ that can be a function of the states of other subsystems in the platoon.

The topology of the closed-loop system is shown in Fig. 1. The integrators at the output of subsystem 1 are appended to the local controller (3) for that subsystem. Note that Fig. 1 does not show the implicit and explicit communications that can exist between subsystems. The feedback signal to the summing junction is the platoon-level function that is either measured by subsystem 1 or communicated by that subsystem from an external source. The exogenous signal entering the summing junction is the reference signal for the platoon-level function. It is either generated exogenously from the platoon or generated by subsystem 1.

The existence of decentralized controllers was studied extensively in works including [8] and [9]. In [6], the idea of decentralized fixed modes was introduced. Essentially, these are modes of the system that cannot be moved by decentralized controllers, as in (3). If a system has no decentralized fixed modes, then a stabilizing decentralized controller such as (3) can be found. An existence test, suitable for our purposes, was presented in [10] based on combining results in [6] and [8]. We adapt that test for the platoon dynamics at hand.

Lemma 3.1 *There exists a decentralized output feedback control system as in (3) that stabilizes the plant (2) with linearized output functions given by $y_i = H_i x$ if*

$$\text{rank} \left[\begin{array}{cc|ccc} \lambda I & -H_c & \hat{G}_{i_1} & \cdots & \hat{G}_{i_\mu} \\ 0 & \lambda I - F & & & \\ \hline & H_{j_1} & & & \\ & \vdots & & & \\ & H_{j_\nu} & & 0 & \end{array} \right] \geq n + p_d \quad (4)$$

is satisfied for all $\lambda \in \mathbb{C}$ and indexes i_1, \dots, i_μ , and j_1, \dots, j_ν , ($0 \leq \mu \leq r$, $0 \leq \nu \leq r$) with

$$\{i_1, \dots, i_\mu\} = \{1, \dots, r\} \setminus \{j_1, \dots, j_\nu\}$$

where \setminus is the set subtraction operator. The hypothesis of Lemma 3.1 need only be checked for values λ that are eigenvalues of F_i and for $\lambda = 0$. An important benefit of our systems-theory approach is that Lemma 3.1 can be used to study the communication structures, both

implicit and explicit, that are required to stabilize the platoon. In general, one would first choose H_i to represent only the implicit communication that is available between subsystems and test the hypothesis of Lemma 3.1. If the rank test succeeds, then no explicit communication is required between subsystems. If the rank test fails, then H_i is altered to represent additional explicit communication channels (e.g., states corresponding to the position of subsystem 2 are explicitly transmitted to subsystem 3). The hypothesis of Lemma 3.1 is checked again, and this iterative processes continues, increasing the explicit communications burden, until the rank test is satisfied.

Due to the structure of the platoon dynamics at hand, more specific conclusions about inter-vehicle communications can be generated. To state these conclusions precisely, we first present some background information.

Definition 3.2 *Subsystem i is locally observable if*

$$\text{rank} \begin{bmatrix} \lambda I - F_i \\ H_{ia} \end{bmatrix} = n_i \quad (5)$$

for all $\lambda \in \mathbb{C}$.

Definition 3.3 *Subsystem i is locally controllable if*

$$\text{rank} \begin{bmatrix} \lambda I - F_i & G_i \end{bmatrix} = n_i \quad (6)$$

for all $\lambda \in \mathbb{C}$.

Definition 3.4 *Subsystem i does not have transmission zero at zero relative to the integrator input if*

$$\text{rank} \begin{bmatrix} -H_{ci} & 0 \\ -F_i & G_i \end{bmatrix} = n_i + \min(p_d, m_i)$$

To efficiently state the theoretical results on existence of decentralized controllers, we list several assumptions.

Assumption 3.5 *Each subsystem is locally controllable and locally observable.*

Assumption 3.6 *The platoon is free of transmission zeros at zero relative to the integrator output. That is, the matrix*

$$\left[\begin{array}{c|ccc} -H_c & & & \\ -F & \hat{G}_1 & \cdots & \hat{G}_r \end{array} \right]$$

has full rank.

Assumption 3.6 states that an integrator can be placed at the output of the integrated function without causing a pole/zero cancellation.

Assumption 3.7 *Every subset of the platoon has no transmission zero at zero relative to the integrator input. That is,*

$$\begin{aligned} \text{rank} & \left[\begin{array}{ccc|ccc} -F_{i_1} & \cdots & 0 & G_{i_1} & & \\ & \ddots & & & \ddots & \\ & & -F_{i_\mu} & & & G_{i_\mu} \\ \hline -H_{ci_1} & \cdots & -H_{ci_\mu} & 0 & \cdots & 0 \end{array} \right] \\ & \geq n_{i_1} + \cdots + n_{i_\mu} + \min(n_p, m_{i_1} + \cdots + m_{i_\mu}) \end{aligned}$$

for each subset

$$\{i_1, \dots, i_\mu\} \subset \{1, \dots, r\}$$

where $0 \leq \mu \leq r$,

Assumption 3.7 states that an integrator can be placed at the output of the integrated function for any subset of the platoon without causing a pole/zero cancellation. Of course, Assumption 3.6 is satisfied if Assumption 3.7 is satisfied. In addition, Assumption 3.7 requires that each subsystem individually is free of zeros at zero with respect to the integrated function output.

We now state a general result for existence of decentralized controllers for the platoon dynamics at hand. Specializations of this result with hypotheses that are more easily verified will be presented in the sequel.

Lemma 3.8 *There exists a decentralized output feedback control system as in (3) that stabilizes the platoon (2) if each subsystem is locally observable (Definition 3.2), if the platoon satisfies Assumption 3.7, and if*

$$\text{rank} \begin{bmatrix} H_{j_1 b_0} \\ \vdots \\ H_{j_\nu b_0} \end{bmatrix} \geq p_d - \min(p_d, m_{j_1} + \cdots + m_{j_\nu}) \quad (7)$$

for all sets of indexes i_1, \dots, i_μ , and j_1, \dots, j_ν , ($\mu \geq 0, \nu \geq 0$) with

$$\{i_1, \dots, i_\mu\} = \{1, \dots, r\} \setminus \{j_1, \dots, j_\nu\}$$

The proof of Lemma 3.8 appears in the Appendix.

It is possible to examine specific communication strategies that satisfy Lemma 3.1. One such communication strategy requires that the integrator states $q(t)$ are broadcast directly to each vehicle in the platoon. In our notation, this is equivalent to choosing $h_{ib}(x(t)) = q(t)$ for all $i = 1, \dots, r$. This choice of explicit communication is desirable in that the bandwidth of the communication is dependant only on the number of integrator states, not on the number of vehicles in the platoon.

Corollary 3.9 *There exists a decentralized output feedback control system as in (3) that stabilizes the platoon (2) if each subsystem is locally observable (Definition 3.2), if the platoon satisfies Assumption 3.7, and if $h_{ib}(x) = q(t)$ for $i = 1, \dots, r$.*

The proof of Corollary 3.9 is omitted since it is a direct extension of the proof of Lemma 3.8. The exact result can also be obtained with hypothesis that is possibly easier to verify.

Corollary 3.10 *There exists a decentralized output feedback control system as in (3) that stabilizes the platoon (2) if Assumptions 3.5 and 3.6 are satisfied and if $h_{ib}(x) = q(t)$ for $i = 1, \dots, r$.*

Corollary 3.10 is not a direct corollary to Lemma 3.8, rather it can be derived from Lemma 3.1. The proof procedes very similarly to that of Lemma 3.8 and is omitted, though we

provide a brief outline. Row and column exchanges of (4) are used to display the matrices in Assumption 3.5, from which the rank of (4) is determined to be at least n . Depending on the selection of $\{i_1, \dots, i_\mu\}$ and $\{j_1, \dots, j_\nu\}$, an additional p_d independent rows or columns are obtained from either Assumption 3.6 or from the hypothesis of the corollary, $h_{ib}(x) = q$.

4 Example

Our decentralized control design method is demonstrated with an example based on controlling a platoon composed of four AUVs. For clarity of presentation, we limit the example to the 2 dimensional plane and choose a simplistic model for the AUV dynamics. Our goal is to control the average and variance of the AUV positions with limited inter-vehicle communications. The dynamics of each vehicle are written

$$\begin{bmatrix} \dot{x}_{1_i} \\ \dot{x}_{2_i} \\ \dot{y}_{1_i} \\ \dot{y}_{2_i} \end{bmatrix} = \begin{bmatrix} x_{2_i} \\ f_i \cos(\theta_i) - 2x_{2_i} \\ y_{2_i} \\ f_i \sin(\theta_i) - 2y_{2_i} \end{bmatrix} \quad (8)$$

where f_i represents the force created by the vehicle's thruster, θ_i is the angle of the AUV in an inertial frame, and the $-2x_{2_i}$ and $-2y_{2_i}$ terms represent viscous damping. The states x_{1_i} and y_{1_i} represent the position of the vehicle within an inertial coordinate frame and x_{2_i} and y_{2_i} are their respective velocities.

It is well known that dynamics such as (8) are nonholonomic. Linearized at a constant operating point, they are not controllable, and thus the vehicles violate the hypothesis of Corollary 3.10. However, the linearized AUV dynamics are controllable when they are linearized about a trajectory. This agrees well with our AUV example, as underwater vehicles are often designed to be in motion.

The AUV dynamics are linearized about a constant velocity given by $\dot{x}_{1_i} = x_{2_i} = \frac{1}{2}$ and $\dot{y}_{1_i} = y_{2_i} = 0$. Inputs corresponding to this trajectory are $f_i = 1$ and $\theta_i = 0$. Selecting a change of variables,

$$\begin{bmatrix} \bar{x}_{1_i} \\ \bar{x}_{2_i} \\ \bar{y}_{1_i} \\ \bar{y}_{2_i} \end{bmatrix} = \begin{bmatrix} x_{1_i} - \frac{1}{2}t \\ x_{2_i} - \frac{1}{2}t \\ y_{1_i} \\ y_{2_i} \end{bmatrix}, \quad \begin{bmatrix} \bar{f}_i \\ \bar{\theta}_i \end{bmatrix} = \begin{bmatrix} f_i - 1 \\ \theta_i \end{bmatrix},$$

where t is time, and then linearizing yields the linear time-invariant system

$$\begin{bmatrix} \dot{\bar{x}}_{1_i} \\ \dot{\bar{x}}_{2_i} \\ \dot{\bar{y}}_{1_i} \\ \dot{\bar{y}}_{2_i} \end{bmatrix} = \begin{bmatrix} 0 & 1 & 0 & 0 \\ 0 & -2 & 0 & 0 \\ 0 & 0 & 0 & 1 \\ 0 & 0 & 0 & -2 \end{bmatrix} \begin{bmatrix} \bar{x}_{1_i} \\ \bar{x}_{2_i} \\ \bar{y}_{1_i} \\ \bar{y}_{2_i} \end{bmatrix} + \begin{bmatrix} 0 & 0 \\ 1 & 0 \\ 0 & 0 \\ 0 & 1 \end{bmatrix} \begin{bmatrix} \bar{f}_i \\ \bar{\theta}_i \end{bmatrix}. \quad (9)$$

Our goal is to regulate the average position of the vehicles and the distribution of the vehicles about the average position in these new coordinates. The average is computed in two dimensions

$$v_x = (\bar{x}_{1_1} + \bar{x}_{1_2} + \bar{x}_{1_3} + \bar{x}_{1_4}) / 4 \quad (10)$$

$$v_y = (\bar{y}_{1_1} + \bar{y}_{1_2} + \bar{y}_{1_3} + \bar{y}_{1_4}) / 4 \quad (11)$$

and the sample variance, which gives the distribution of the vehicles, is also computed in two dimensions

$$w_x = \sum_{i=1}^4 (\bar{x}_{1_i} - v_x)^2 / 3 \quad (12)$$

$$w_y = \sum_{i=1}^4 (\bar{y}_{1_i} - v_y)^2 / 3. \quad (13)$$

The global (platoon-level) functions (10)–(13) are entries of the vector function $h_c(\bar{x}) = [v_x, v_y, w_x, w_y]^T$ and are integrated to ensure zero steady-state error. Measurement and integration of the global variables is accomplished by subsystem 1, which we consider the *mothership* AUV. Subsystem 1 may have the capability to directly measure the global variables, or to indirectly measure the global variables by measuring the position of every other

vehicle in the platoon and computing (10) – (13). Importantly, subsystem 1 is the only subsystem with this capability.

To partially satisfy the hypothesis of Corollary 3.10, the integrated global variables are broadcast to the platoon. This is equivalent to setting $h_{ib}(x(t)) = q(t)$, $i = 2, 3, 4$. In addition, we assume that each vehicle can measure its own position so that each subsystem is locally observable. The platoon is linearized at $(\bar{x}_{11}, \bar{y}_{11}) = (1.1, 2)$, $(\bar{x}_{12}, \bar{y}_{12}) = (2, 1.5)$, $(\bar{x}_{13}, \bar{y}_{13}) = (0.7, 1.1)$, and $(\bar{x}_{14}, \bar{y}_{14}) = (1.5, 0.7)$. Inspection of the linearized subsystem dynamics (9) shows that each subsystem is locally controllable and that the platoon is free of zeros at zero. Thus Assumptions 3.5 and 3.6 are satisfied, which implies that the hypothesis of Corollary 3.10 is satisfied and a controller (3) for each subsystem exists such that the platoon as a whole is stable and is able to track global function commands. The requirements for such a decentralized control system are that one vehicle can measure and integrate the global variables and that the integrated global variables are broadcast to the other vehicles in the platoon. The only communication that is required is the broadcast of the integrated global variables.

4.1 Synthesis of Decentralized Controllers

From Corollary 3.10, a decentralized controller exists for the platoon operating near the stated equilibrium point. In general, however, designing such a controller is a difficult task. For the system at hand, we note that the output of each system, $y_i(t) = H_i x(t)$ is not observable. Thus a number of design methods reported in the literature are not applicable. Instead, we adapt a synthesis technique recently reported in [11] to design a decentralized controller for the platoon at hand. Our focus here is on existence of decentralized controllers with attention to communication requirements. Thus we refer the reader to [11], and references therein, for a detailed discussion of decentralized controller synthesis and present only a brief summary of the procedure here.

Let the decentralized controller (3) coefficient matrices be described by A , B , C , where

$$A = \begin{bmatrix} A_1 & 0 & \cdots & 0 \\ 0 & A_2 & & \\ \vdots & & \ddots & \\ 0 & & & A_r \end{bmatrix} \in \mathbb{R}^{n_k \times n_k},$$

$n_k = n_{k1} + \cdots + n_{kr}$, and B and C are described similarly. We also define the matrices

$$F_p = \left[\begin{array}{c|c} H_c & 0 \\ \hline F & \\ \hline 0 & I_{n_k} \end{array} \right], \quad G_p = \left[\begin{array}{ccc|c} \hat{G}_1 & \cdots & \hat{G}_r & 0 \\ \hline & 0 & & I_{n_k} \end{array} \right], \quad H_p = \left[\begin{array}{c|c} H_1 & 0 \\ \vdots & \\ \hline H_r & \\ \hline 0 & I_{n_k} \end{array} \right]$$

and

$$K = \begin{bmatrix} 0 & C \\ B & A \end{bmatrix}$$

Note that the coefficient matrix for the closed-loop state dynamics (zero input) is $F_p + G_p K H_p$. A decentralized controller is found by solving the linear matrix inequality (LMI)

$$\begin{bmatrix} \epsilon Q - R & (\epsilon F_p + G_p \bar{K} H_p)^T \\ \epsilon F_p + G_p \bar{K} H_p & -W \end{bmatrix} < 0 \quad (14)$$

with the constraint

$$W = R^{-1}. \quad (15)$$

Where $\bar{K} = K\epsilon$ has same structure as K . The free variables are the matrix \bar{K} , symmetric positive definite matrices R , W , and positive constant ϵ . The symmetric positive-definite matrix Q is selected as a design parameter and is constant in (14).

Synthesis problems where structural constraints have been imposed on the controller, such as fixed order control or decentralized control, do not have desirable convexity properties to aid in computing a solution. Indeed, (14) is convex in the unknown variables but the

constraint (15) is not. Nonetheless, an algorithm composed of alternating projections is described in [11] to obtain solutions that satisfy (14) and (15) simultaneously. Though the global convergence of the alternating projection algorithm cannot be obtained, local convergence can. The algorithm proceeds as follows: \bar{K} , R , W , ϵ are found that satisfy the LMI (14), then projected onto the space of matrices for which $W = R^{-1}$. The new projected values of R and W are then projected onto the space of matrices that solve (14). This process continues until (14) and (15) are simultaneously satisfied. Numerical methods for this algorithm are found in [11].

We found the four subsystem controllers with controller state dimension $n_{k1} = \dots = n_{k4} = 15$, positive constant ϵ , and symmetric positive-definite matrices R and W that simultaneously satisfy (14) and (15) when $Q = I$. These controllers are able to stabilize the closed-loop system in the neighborhood of where the system was linearized with adequate performance. The controller matrix coefficients A_{k1}, \dots, A_{k4} , B_{k1}, \dots, B_{k4} , and C_{k1}, \dots, C_{k4} are listed in the second Appendix. The algorithm is numerically intensive which causes the controller design process to be prohibitively time consuming. Though we have proved existence of decentralized controllers for the vehicle platoon, efficient methods of control synthesis remain an open issue.

4.2 Simulation

The AUV platoon was simulated using nonlinear global (platoon-level) variable functions, nonlinear dynamics, and linear decentralized controllers. Commanded and actual trajectories for the average and variance along the x - and y -axis are shown in Fig. 2 and 3, respectively. Along the x -axis, the data is shown with respect to the transformed state variables \bar{x}_{1i} , for convenience. The trajectories of the vehicles are shown in Fig. 4 and close-up snapshots of the vehicles over time are shown in Fig. 5.

Note that the vehicle spacing about the mean position is not regular or specified. There are an infinite number of possible full system states that satisfy the required regulation of the global variables. This allows for much more flexible system behavior than that of a platoon

under rigid formation control while still guaranteeing that the desired objectives are met.

5 Concluding Remarks

In this article a decentralized control design methodology for regulating global functions of cooperating mobile systems has been presented. Application of relatively standard system-theoretic tools, such as decentralized control, leads to a novel broadcast-only communication structure (single-source, unidirectional). The feedback mechanism between vehicles is the measurement of the global variables (by a single unit) and broadcast of their integrated values. More generally, methods presented here allow the designer to determine what explicit communication strategies are sufficient for a stabilizing decentralized control to exist.

Using a simplified model, we have shown that it is indeed possible to regulate global variables of a platoon of autonomous underwater vehicles: in particular, the center of the platoon and the distribution of the vehicles about the center. Significant features of the developed system are that a relatively small amount of explicit communication is required between vehicles and that no vehicle must regulate its actual position. Further, the approach presented is scalable to any number of cooperating vehicles without the need for additional communication, although there is a practical limit on the size of the platoon.

This work has application in a variety of domains other than AUV platoons; limitation of active bandwidth is desirable in cases of reconnaissance, limited power applications, and very-large-scale platoons. A number of open research areas remain, including analysis of the performance of the system under disturbances, failure of a subsystem, and efficient methods for synthesizing decentralized controllers.

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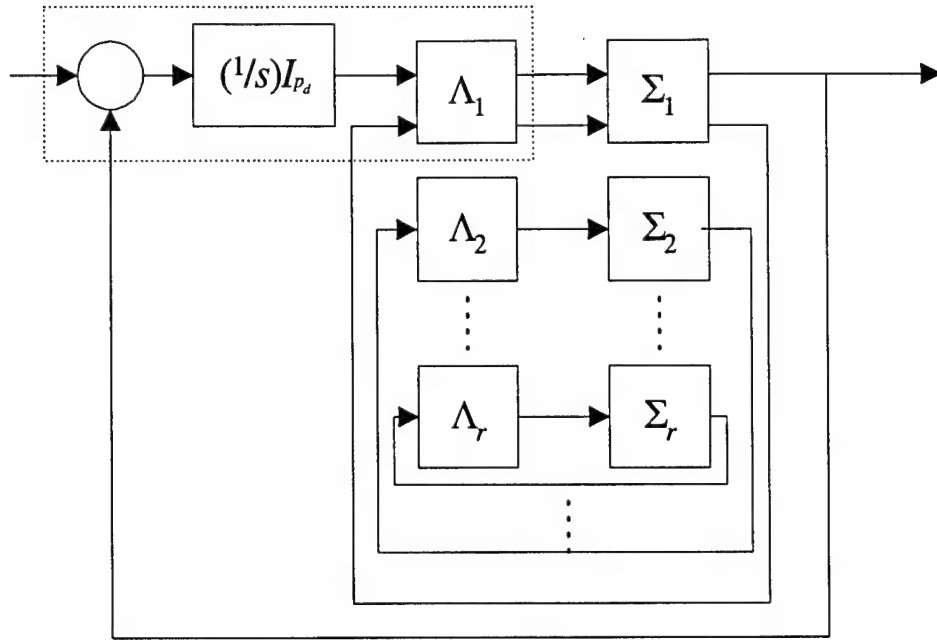


Figure 1: Block diagram of closed-loop system. Subsystems and local controllers are denoted Σ and Λ , respectively. The dashed box signifies that the controller for subsystem 1 incorporates the integrators.

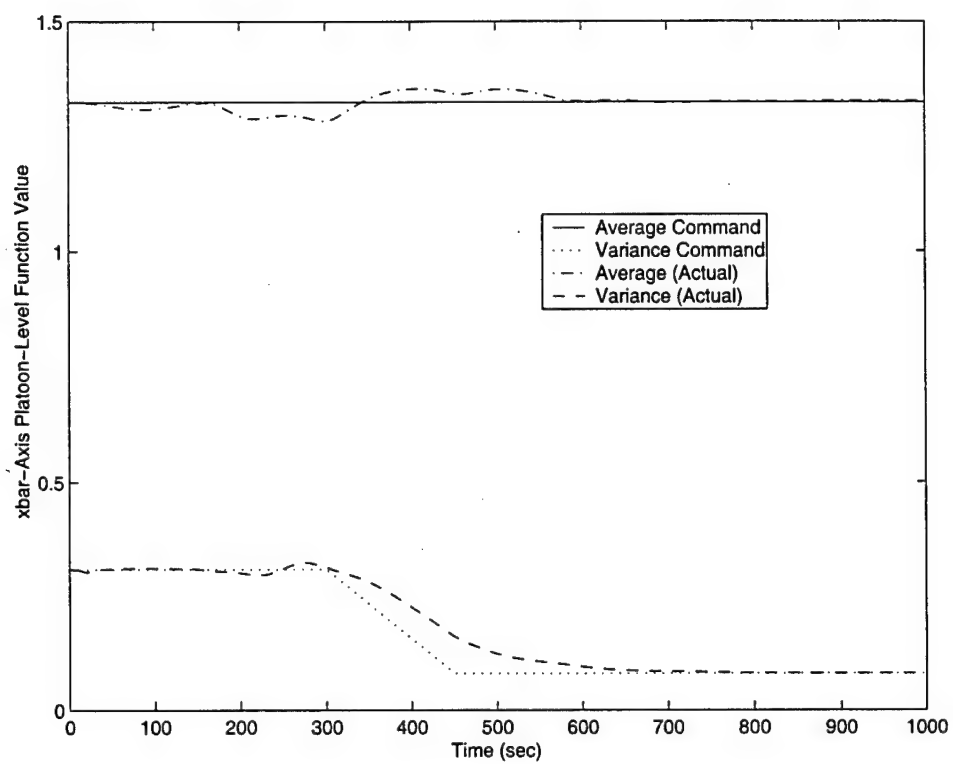


Figure 2: Actual and commanded average position and variance for \bar{x} .

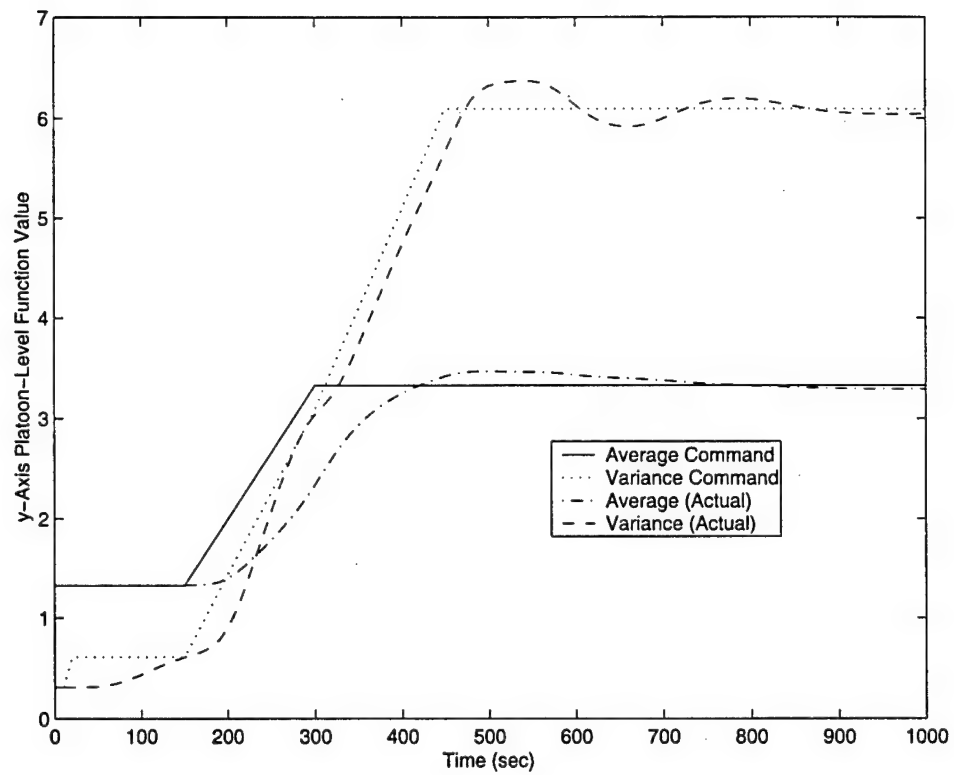


Figure 3: Actual and commanded average position and variance for y.

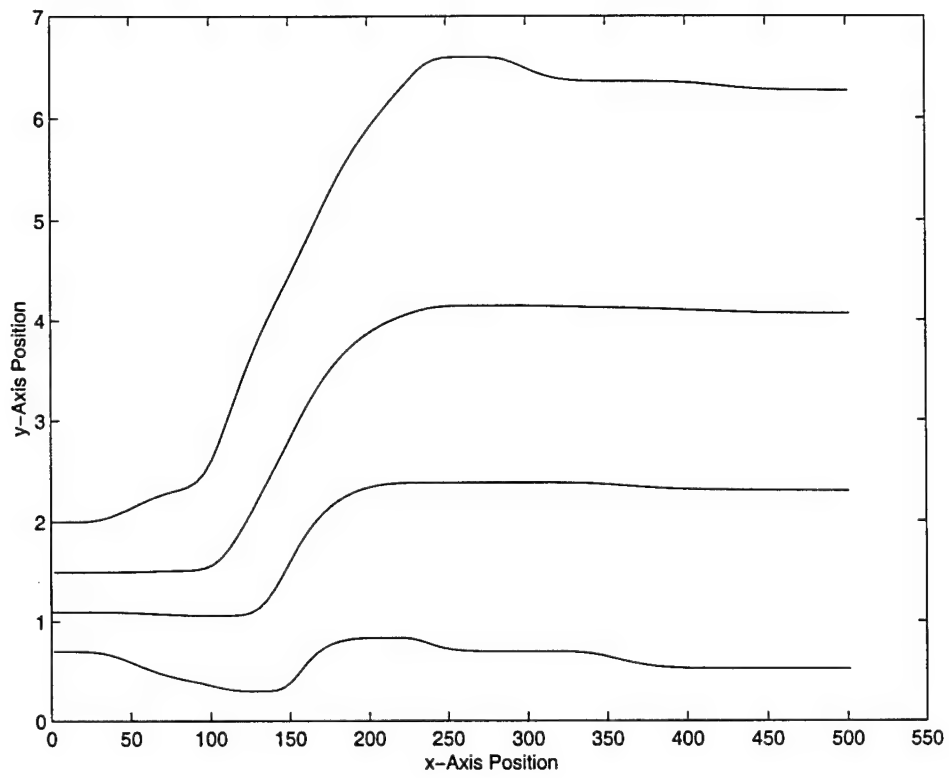


Figure 4: Trajectories of vehicles.

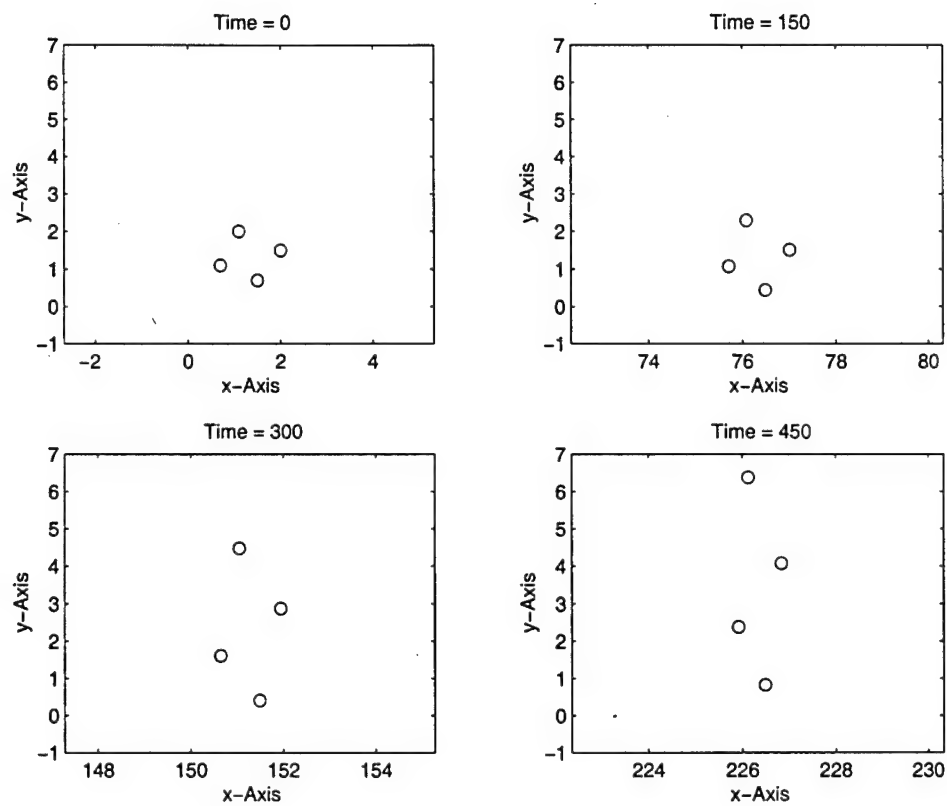


Figure 5: Snapshots of vehicle positions over time.

6 Appendix – Proofs

Proof of Lemma 3.8:

Our proof is based on the hypothesis of Lemma 1. Define

$$Q = \left[\begin{array}{cc|ccc} \lambda I & -H_c & 0 & \cdots & 0 \\ 0 & \lambda I - F & \bar{G}_{i_1} & \cdots & \bar{G}_{i_\mu} \\ \hline H_{j_1} & & & & \\ \vdots & & 0 & & \\ H_{j_\nu} & & & & \end{array} \right] \quad (16)$$

Our goal is to show that under the hypothesis of Lemma 2, $\text{rank } Q \geq n + p_d$, the platoon state dimension. The entries of Q are expanded to display

$$Q = \left[\begin{array}{cccc|cccc} \lambda I & -H_{c1} & \cdots & -H_{cr} & 0 & \cdots & 0 \\ 0 & \lambda I - F_1 & \cdots & 0 & G_{i_1} & & \\ \vdots & & \ddots & & & \ddots & \\ 0 & & & \lambda I - F_r & & & G_{i_r} \\ \hline 0 & H_{j_1a} & \cdots & 0 & & & \\ H_{j_1b_0} & H_{j_1b_1} & \cdots & H_{j_1b_r} & & & \\ \vdots & & & \vdots & & & 0 \\ 0 & \cdots & 0 & H_{j_\nu a} & & & \\ H_{j_\nu b_0} & H_{j_\nu b_1} & \cdots & H_{j_\nu b_r} & & & \end{array} \right] \quad (17)$$

and then rows and columns are exchanged, yielding

$$\bar{Q} = \left[\begin{array}{c|ccc|ccc|ccc} \lambda I & -H_{ci_1} & \cdots & -H_{ci_\mu} & -H_{cj_1} & \cdots & -H_{cj_\nu} & 0 & \cdots & 0 \\ \hline 0 & \lambda I - F_{i_1} & & & 0 & & & G_{i_1} & & \\ \vdots & & \ddots & & & \ddots & & & \ddots & \\ 0 & & & \lambda I - F_{i_\mu} & & & 0 & & & G_{i_\mu} \\ \hline 0 & 0 & & & \lambda I - F_{j_1} & & & 0 & & \\ \vdots & & \ddots & & & \ddots & & & \ddots & \\ 0 & & & 0 & & & \lambda I - F_{j_\nu} & & & 0 \\ \hline H_{j_1 b_0} & H_{j_1 b_{i_1}} & \cdots & H_{j_1 b_{i_\mu}} & H_{j_1 b_{j_1}} & \cdots & H_{j_1 b_{j_\nu}} & 0 & & \\ \vdots & \vdots & \ddots & & \vdots & \ddots & & & \ddots & \\ H_{j_\nu b_0} & H_{j_\nu b_{i_1}} & & H_{j_\nu b_{i_\mu}} & H_{j_\nu b_{j_1}} & & H_{j_\nu b_{j_\nu}} & & & 0 \\ \hline 0 & 0 & & & H_{j_1 a} & & & 0 & & \\ \vdots & & \ddots & & & \ddots & & & \ddots & \\ 0 & & & 0 & & & H_{j_\nu a} & & & 0 \end{array} \right]$$

Futher exchanging rows and columns yields,

$$\tilde{Q} = \left[\begin{array}{c|cccc|cccc} \lambda I & H_{ci_1} & 0 & \cdots & H_{ci_\nu} & 0 & H_{cj_1} & \cdots & H_{cj_\nu} \\ \hline 0 & \lambda I - F_{i_1} & G_{i_1} & & & & 0 & & \\ \vdots & & & \ddots & \ddots & & & \ddots & \\ 0 & & & & \lambda I - F_{i_\nu} & G_{i_\nu} & & & 0 \\ \hline 0 & 0 & & & & & \lambda I - F_{j_1} & & \\ \vdots & & \ddots & & & & H_{j_1 a} & \ddots & \\ & & & & & & & \ddots & \\ \vdots & & & & & & & & \lambda I - F_{j_\nu} \\ 0 & & & & & 0 & & & H_{j_\nu a} \\ \hline H_{j_1 b_0} & H_{i_1 b_1} & 0 & \cdots & H_{i_1 b_{i_\nu}} & 0 & H_{j_1 b_0} & \cdots & H_{j_\nu b_{i_\nu}} \\ \vdots & \vdots & \vdots & & \vdots & \vdots & \vdots & & \vdots \\ H_{j_\nu b_0} & H_{i_\mu b_1} & 0 & \cdots & H_{i_\mu b_{i_\nu}} & 0 & H_{j_\nu b_0} & \cdots & H_{\nu_\nu b_{i_\nu}} \end{array} \right]$$

Of course row and column exchanges do not alter matrix rank, so

$$\text{rank } Q = \text{rank } \bar{Q} = \text{rank } \tilde{Q}$$

Assumption 3.7 implies that the first the first block partitioned row of \tilde{Q} has rank greater than or equal to $n_{i_1} + \dots + n_{i_\mu} + \min(n_p, m_{i_1} + \dots + m_{i_\mu})$. By the subsystem observability condition (5), the second block partitioned row has rank $n_{j_1} + \dots + n_{j_\nu}$. And by the structure of \tilde{Q} , it can be seen that the first and second block partitioned rows of \tilde{Q} are linearly independent. Thus the first and second block partitioned rows have rank $n_1 + \dots + n_r + \min(n_p, m_{i_1} + \dots + m_{i_\mu})$. If $\lambda \neq 0$, then the first block partitioned row in fact has rank $p_d + n_{i_1} + \dots + n_{i_\nu}$, and the proof is complete. For $\lambda = 0$, we note that the (3, 1) partition of \tilde{Q} is linearly independent from the first two block partitioned rows. Thus

$$\text{rank } \tilde{Q} \geq \text{rank} \begin{bmatrix} H_{j_1 b_0} \\ \vdots \\ H_{j_\nu b_0} \end{bmatrix} + n_1 + \dots + n_r + \min(p_d, m_{j_1} + \dots + m_{j_\nu})$$

and by hypothesis,

$$\text{rank} \begin{bmatrix} H_{j_1 b_0} \\ \vdots \\ H_{j_\nu b_0} \end{bmatrix} \geq p_d - \min(p_d, m_{j_1} + \dots + m_{j_\nu}) \quad (18)$$

Thus $\text{rank } \tilde{Q} \geq n_1 + \dots + n_r + p_d = n + p_d$ and decentralized output feedback controller exists.

■

7 Appendix – Controller matrix coefficients

$A_{k1} =$

-1.0874	-0.5699	-0.3583	-0.1502	-0.2465	-0.7191	-0.4427	-0.0976	-0.0101	-0.3640	-0.4205	-0.0416	0.0070	-0.0771	-0.0865
-1.2646	-1.4790	-0.3914	0.4453	0.3415	0.4448	-0.0781	-0.0303	0.0315	0.2285	0.0865	-0.1090	-0.1019	0.2252	-0.0696
0.1633	0.9085	-1.0076	-0.2136	0.6534	0.4345	-0.2953	0.1169	0.4783	0.1961	0.1260	0.1378	-0.0935	-0.1473	-0.0005
0.0430	0.0417	0.5654	-2.1293	0.0369	0.2845	-0.1317	-0.0798	0.3475	-0.0652	0.1052	-0.0282	-0.1806	-0.2470	0.2603
-0.2238	-0.2171	0.0591	-0.7110	-0.9275	0.0651	0.0392	0.2987	-0.3053	0.0197	0.1266	0.3043	0.6329	0.4582	-0.2918
0.3182	0.3086	-0.0840	0.0317	-0.2384	-1.0643	-0.2429	0.0910	0.4523	-0.2470	-0.2285	-0.1970	-0.4746	0.2039	0.5781
0.3445	0.3341	-0.0910	0.0343	0.4169	-0.0378	-1.4139	-0.1016	-0.4081	-0.4484	0.3220	0.1844	0.1756	0.6717	0.2178
0.2234	0.2167	-0.0590	0.0223	0.2703	-0.1992	1.0513	-2.0775	0.4775	-0.3483	0.5507	-0.1435	0.2732	0.1327	-0.1793
-0.2462	-0.2388	0.0650	-0.0245	-0.2980	0.2195	0.3561	-0.9160	-1.4319	-0.0586	0.3847	-0.0039	-0.7203	-0.3762	0.1081
0.2226	0.2160	-0.0588	0.0222	0.2694	-0.1985	-0.3220	-0.1008	-0.2854	-0.8585	-0.1026	0.6194	-0.0385	-0.8426	-0.6091
0.0770	0.0747	-0.0203	0.0077	0.0932	-0.0686	-0.1113	-0.0348	0.0865	0.3470	-1.1686	-0.4992	0.2421	-0.5115	-0.2286
0.0169	0.0164	-0.0045	0.0017	0.0205	-0.0151	-0.0245	-0.0077	0.0190	-0.0174	0.3465	-1.6882	0.3018	-0.4887	0.0184
0.0078	0.0075	-0.0021	0.0008	0.0095	-0.0070	-0.0113	-0.0036	0.0088	-0.0081	0.0025	0.6283	-1.4259	0.0510	-0.1521
0.0021	0.0020	-0.0005	0.0002	0.0025	-0.0018	-0.0030	-0.0009	0.0023	-0.0021	0.0007	-0.0015	0.2255	-0.9198	0.0222
0.0000	0.0000	-0.0001	-0.0000	0.0001	-0.0001	-0.0001	-0.0001	0.0001	-0.0001	-0.0000	-0.0000	-0.0000	0.0859	-1.0491

$A_{k2} =$

-1.7171	-0.7896	-0.0086	0.0772	-0.0247	-0.0431	-0.1354	0.2373	-0.0150	-0.2735	-0.5332	0.1931	0.0033	0.2121	0.7672
-0.0098	-1.5755	0.2993	0.0375	0.0404	-0.0100	-0.0023	0.0218	-0.0581	0.0596	0.0172	-0.0600	0.0258	0.0758	0.2916
0.4242	-0.7744	-1.2703	-0.2526	0.1091	0.0809	0.0298	-0.1229	0.1726	-0.3040	-0.1333	0.3143	-0.0911	0.4173	-0.9307
-0.0631	-0.2358	-0.1828	-1.5647	0.6090	0.0079	0.0378	-0.1194	0.0652	0.2241	0.4412	-0.1617	-0.0091	-0.3256	-0.7155
0.1976	0.0607	0.0327	0.7019	-1.0395	0.3414	-0.1952	0.2567	0.0924	-0.1870	-0.3060	0.1408	-0.0159	0.0097	0.2188
0.0897	-0.0258	0.0029	0.1803	-0.3601	-1.6413	0.9893	0.2769	0.1088	-0.0685	-0.0834	0.0682	-0.0138	0.0659	-0.0567
0.0291	0.1221	0.1623	-0.2089	0.0183	-0.5616	-2.0914	-0.4716	0.1133	-0.0869	-0.1921	0.0684	-0.0020	0.0067	0.2356
0.1257	-0.1613	0.0165	0.5545	0.6824	-0.1837	-0.3232	-1.1821	0.1375	-0.0764	-0.0179	0.0559	-0.0036	0.3427	-0.0153
0.1587	-0.0865	-0.0004	-0.4520	-0.4762	0.2059	0.0056	-0.4844	-1.2374	0.0439	-0.0657	0.1302	-0.0182	0.5698	-0.1322
-0.4832	-0.0378	0.2108	0.1061	-0.0293	-0.0735	0.3322	-0.1972	-0.1950	-1.3514	0.2255	-0.2816	0.0675	0.3061	0.2012
-0.7056	0.0449	-0.0435	0.4092	0.7438	0.1407	-0.0202	-0.1688	-0.1338	0.1230	-0.7493	-0.2518	0.1024	0.1734	0.2202
0.5281	-0.1038	-0.0502	0.0829	0.4201	-0.1144	-0.3296	0.3601	0.1531	-0.9494	0.0832	-1.4590	0.0433	-0.1954	-0.2983
-0.1307	0.0970	0.2822	-0.3473	-0.0751	-0.2920	-0.3857	-0.4097	-0.0869	-0.7491	0.1585	-0.8790	-1.3383	0.1585	0.1087
0.2860	0.0110	0.2097	0.6164	0.0570	0.0519	0.2293	-0.6001	-0.1885	0.0489	0.0945	0.0811	0.0027	-0.2128	-0.0379
-0.7106	0.1239	-0.0429	-0.0977	-0.1206	0.0365	-0.0162	-0.0303	0.0878	-0.2737	-0.2889	0.2369	-0.0279	-0.1514	-1.3984

$A_{k3} =$

-0.9105	-0.2563	0.0057	-0.0534	-0.1463	0.2496	0.0805	-0.2868	0.8434	0.8237	0.4871	0.1814	-0.2238	0.0380	-0.2194
-0.8017	-1.7919	-0.5188	-0.4718	0.2395	0.1128	0.3361	-0.0196	-0.3226	-0.2341	-0.1348	-0.1529	0.1925	0.2854	0.2569
0.0122	0.7669	-1.1826	-0.0555	-0.4111	-0.3321	0.3741	0.0278	-0.1851	-0.0948	0.0699	0.1139	-0.4690	0.7205	-0.0552
-0.0046	-0.0567	-0.4850	-2.0307	0.0393	0.4830	-0.1613	-0.0375	0.1601	0.0784	-0.1332	-0.1742	0.4124	-0.3091	-0.2517
0.0048	0.0599	-0.0151	-0.8874	-1.1645	0.0748	-0.1107	0.1123	0.0608	0.0242	0.2273	0.3092	-0.6239	0.1473	0.4118
0.0073	0.0908	-0.0230	0.0051	-0.5958	-0.9641	0.0670	-0.1211	-0.0699	0.1243	-0.2418	-0.3630	0.6387	0.2145	-0.8157
-0.0374	-0.4651	0.1177	-0.0260	0.2662	0.6043	-1.0590	-0.2316	0.0655	-0.3699	0.0039	0.1199	-0.3668	0.9929	-0.6523
0.0300	0.3731	-0.0944	0.0208	-0.2135	-0.1679	-0.3058	-1.9320	0.7928	-0.9375	-0.6360	-0.1214	-0.2732	-0.2348	0.1522
-0.0513	-0.6381	0.1615	-0.0356	0.3652	0.2871	-0.2302	-0.4551	-1.3740	0.1192	-0.0090	-0.0379	-0.2743	-0.1509	0.0765
-0.0410	-0.5106	0.1292	-0.0285	0.2923	0.2297	-0.1842	0.2255	0.7190	-1.6842	-0.1975	0.4060	0.0578	-0.0347	0.0867
-0.0210	-0.2607	0.0660	-0.0146	0.1492	0.1173	-0.0940	0.1151	-0.2013	0.4258	-1.0882	-0.5723	-0.3127	-0.0646	0.0931
-0.0050	-0.0627	0.0159	-0.0035	0.0359	0.0282	-0.0226	0.0277	-0.0484	-0.0394	0.3996	-1.8865	-0.3577	-0.0944	0.0513
0.0032	0.0393	-0.0099	0.0022	-0.0225	-0.0177	0.0141	-0.0173	0.0302	0.0246	0.0056	-0.7655	-1.2264	-0.0370	-0.1348
-0.0014	-0.0179	0.0045	-0.0010	0.0102	0.0081	-0.0065	0.0079	-0.0139	-0.0113	-0.0026	0.0051	-0.3754	-0.8237	0.0099
0.0002	0.0027	-0.0006	0.0001	-0.0015	-0.0012	0.0009	-0.0011	0.0019	0.0016	0.0003	-0.0008	0.0016	-0.1264	-0.9702

$A_{k4} =$

-1.6158	0.7019	0.0236	-0.0886	0.0575	-0.0390	-0.0961	0.1011	0.0286	-0.0961	0.0350	-0.1659	-0.3826	0.3785	0.2096
0.3856	-0.7928	0.3508	-0.0211	-0.0064	-0.0305	-0.0548	0.0343	0.0075	-0.0213	-0.0142	-0.0014	-0.1894	0.0555	-0.1830
0.4092	-0.6513	-1.6856	-0.7982	-0.0199	-0.0923	-0.1471	0.1329	0.0315	-0.0637	0.0415	-0.1344	-0.1953	0.3465	0.2686
0.4652	0.1228	-0.2471	-1.1896	-0.2294	0.1063	0.1030	-0.0145	-0.0198	-0.1204	0.0589	-0.1915	-0.5154	0.7393	0.1085
0.5135	0.4796	-0.0811	0.0697	-1.2852	0.2072	-0.2502	0.1674	0.0629	-0.0682	-0.0301	-0.0702	-0.4166	-0.0805	-0.0979
-0.3277	-0.4398	-0.0857	0.0284	-0.8860	-1.6910	1.1527	0.2368	0.0710	-0.0191	-0.0108	-0.0351	-0.0997	-0.0864	0.0631
0.0616	-0.1034	-0.3639	0.5828	0.0918	0.1320	-1.4844	-0.6008	0.0903	-0.0802	-0.0071	-0.1178	-0.4071	0.0694	0.0653
0.3472	0.4165	0.1031	-0.1450	-0.0747	-0.4560	0.2271	-1.3913	0.2516	-0.0194	0.0080	-0.0125	-0.1287	0.2036	-0.1150
0.1285	0.2384	-0.2758	0.7719	-0.3086	0.3826	0.5213	-0.2187	-1.2364	0.1203	0.0141	-0.0562	-0.2959	0.3814	-0.1378
0.0304	0.0307	-0.3653	0.2764	0.5737	0.1363	-0.0604	-0.0309	-0.4386	-1.9970	-0.9017	0.0296	0.2349	0.2560	0.0497
0.2146	0.4918	0.1874	-0.2680	0.2052	-0.5072	0.0428	0.5441	-0.2531	0.0835	-1.4919	-0.1728	-0.2477	-0.1582	-0.1142
0.2745	0.3443	-0.0440	-0.1661	-0.1647	-0.3422	0.2594	-0.0867	0.1175	0.1984	0.0855	-1.1733	0.0382	0.1002	0.2040
-0.3949	-0.1526	0.2620	-0.5824	0.3802	0.1240	0.2283	-0.0190	-0.0050	0.1165	0.2703	-0.0497	-1.1250	1.0262	-0.0044
0.1250	0.2418	0.0777	-0.2534	0.2322	0.2212	0.5231	-0.3780	-0.1404	0.1152	0.0432	0.2311	0.4460	-0.6716	-0.5291
-0.7093	-0.3022	-0.0130	-0.1629	0.0828	-0.1329	-0.3035	0.1708	0.0745	-0.2175	-0.0531	-0.1991	-0.9868	0.3949	-1.3133

$B_{k1} =$

-0.5379	0.1284	0.1909	-0.3909	0.6741	-0.0613
0.4538	-0.1068	-0.1292	0.4210	-0.1290	-0.0909
1.0584	-0.1306	0.1099	0.4186	0.7198	-0.1393
0.3407	0.2512	-0.0572	0.1386	0.0377	0.0282
-0.3761	-0.8547	-0.0882	-0.1598	0.1486	-0.0806
-0.9366	0.7001	-0.1831	0.1204	0.1534	-0.3335
-0.2538	-0.5205	-0.4143	0.2871	0.1375	0.6287
-0.2251	-0.3458	-0.2376	0.2489	0.1921	0.0741
0.0535	0.6031	0.2188	-0.0314	0.0943	0.4561
0.1932	-0.1535	0.6238	-1.0852	-0.0662	-0.1201
0.1516	0.4098	0.3048	-0.5291	0.0478	0.5369
0.1722	0.4762	-0.4000	-0.1682	0.1486	0.1595
0.2570	0.7209	-0.2612	-0.0570	0.0964	0.0217
0.2000	0.1060	0.6521	-0.3042	-0.1054	-0.2754
0.3019	-0.3387	-0.4854	-0.7357	0.1788	-0.2075

$B_{k2} =$

-0.2909	0.0073	0.1903	0.0084	0.2058	-0.0123
0.0805	-0.0078	-0.5242	0.2839	0.0117	0.0378
-0.0062	0.0026	-1.0379	0.8958	-0.2814	0.1817
-0.1327	-0.0369	0.5580	-0.1763	-0.1191	-0.1984
-0.4544	-0.1188	-0.4400	0.3242	0.1701	0.1421
-0.1994	-0.4418	-0.2826	-0.1656	-0.0453	-0.5776
-0.5704	-0.1832	-0.2058	0.0147	-0.0231	-0.3001
1.1469	0.1252	-0.1883	0.4370	0.2481	-0.2379
0.3164	-1.1987	-0.6632	-0.7297	-0.1308	0.0141
0.3416	-0.3368	0.2187	0.3422	-0.0315	-0.3992
0.5252	0.6319	-0.2209	-0.7778	-0.2569	0.1987
-0.1911	0.8272	-0.6375	-0.3375	-0.0166	-0.0223
0.0344	-0.5583	0.3345	0.1205	0.0534	0.6368
0.7842	-0.0530	-0.1212	0.0793	0.3532	0.0304
-0.1597	0.0191	-0.0839	-0.4120	0.8788	0.0253

$B_{k3} =$

0.6552	-0.0054	-0.8726	0.3643	-0.5112	0.0153
-0.4078	-0.0710	0.1845	-0.2122	-0.0633	-0.0289
-1.3601	-0.0567	-0.0766	-0.1446	-0.5907	-0.0390
0.4605	-0.2895	0.3048	-0.1611	-0.0514	0.0408
-0.3243	0.6441	-0.3600	0.4180	0.6141	0.0469
-0.0349	-1.1244	0.1712	-0.6085	0.2598	0.1103
-0.3859	-0.7037	-1.0784	0.5244	0.2742	0.0467
0.1252	0.0445	0.3457	-0.1209	-0.1637	0.3802
-0.2085	0.2996	-0.0908	-0.1798	0.0589	0.6537
-0.1357	0.2537	0.0762	-0.2373	0.0539	-0.2104
0.0977	-0.1584	0.1861	0.1187	0.0599	-0.7033
0.0897	-0.2004	0.2695	0.1205	-0.0458	0.0231
-0.1905	0.3074	-0.4844	-0.0883	0.1067	-0.1470
0.3362	0.5740	-0.1024	-0.4192	0.0439	0.1410
0.1889	-0.4359	-0.4948	-0.6071	0.1144	-0.0158

$B_{k4} =$

0.0525	0.0170	0.0534	-0.0410	-0.0143	-0.0754
0.1176	0.0306	-0.1188	0.1378	0.1784	-0.1642
0.1742	-0.0118	-0.2564	0.5309	-0.2681	-0.2630
-0.6412	0.0732	0.3290	-0.2807	-0.0632	0.2887
0.5304	-0.1225	0.6373	-0.9498	-0.1060	0.5963
0.7030	-0.4348	-0.1298	0.3775	0.0586	-0.1475
0.7997	-0.1025	0.2524	-0.2841	0.0802	-0.2595
-0.5719	-0.5199	0.3961	-0.3436	-0.1595	-0.4754
-0.0543	-1.4304	0.1311	-0.4135	-0.0424	-0.1838
-0.1310	-0.2114	-0.3112	-0.1828	0.1264	0.2672
-0.0671	0.6981	0.9114	-0.2930	-0.0829	-0.4076
-0.1609	0.2528	-0.8392	-1.0804	0.2796	-0.2420
-0.5765	-0.2115	0.1948	0.2573	0.3006	0.0617
-0.7573	-0.0433	-0.1656	0.0543	-0.5216	0.0397
0.2588	0.0946	-0.4482	-0.2142	-0.7473	0.0092

$$C_{k1} = \begin{bmatrix} 0.7396 & 0.2995 & 0.6316 & -0.1767 & 0.0704 & 0.1805 & 0.1500 & 0.2008 & 0.1113 & 0.0077 & 0.1040 & 0.1110 & 0.0592 & -0.0842 & 0.1361 \\ -0.0615 & -0.0431 & -0.0526 & 0.0669 & -0.0823 & -0.1573 & 0.8049 & -0.3515 & 0.5074 & -0.1004 & 0.3799 & 0.0137 & -0.0608 & -0.1739 & -0.1492 \end{bmatrix}$$

$$C_{k2} = \begin{bmatrix} -0.0596 & -0.0605 & -0.0114 & 0.1545 & 0.0704 & -0.0434 & -0.0657 & 0.1467 & -0.1613 & 0.0880 & -0.1342 & -0.1017 & 0.0653 & 0.2722 & 1.0390 \\ 0.0031 & 0.0009 & 0.1616 & -0.1672 & 0.3112 & -0.6263 & -0.5153 & -0.2603 & 0.0762 & -0.4500 & 0.1576 & -0.1987 & 0.2828 & 0.0009 & 0.0166 \end{bmatrix}$$

$$C_{k3} = \begin{bmatrix} -0.4916 & -0.4995 & -0.4794 & -0.2173 & 0.5268 & 0.3399 & 0.2128 & -0.1526 & 0.0924 & 0.1077 & 0.0865 & -0.0352 & 0.0748 & 0.0602 & 0.0881 \\ -0.0111 & -0.0261 & -0.0397 & 0.0021 & 0.0574 & 0.0570 & -0.0424 & 0.1918 & 0.7449 & -0.5332 & -0.5499 & 0.1448 & -0.1388 & 0.0510 & 0.0032 \end{bmatrix}$$

$$C_{k4} = \begin{bmatrix} 0.1145 & 0.1341 & -0.0726 & 0.1653 & -0.1230 & 0.0444 & 0.1149 & -0.1391 & -0.0430 & 0.1370 & -0.0966 & 0.2809 & 0.4040 & -0.7081 & -0.5853 \\ 0.0113 & 0.0116 & -0.2330 & 0.3201 & 0.6736 & -0.0083 & -0.5086 & -0.4356 & -0.1654 & -0.0248 & -0.4083 & -0.1443 & 0.0250 & -0.0003 & 0.0076 \end{bmatrix}$$

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CHAPTER VI

FIGURE 5

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1. AGENCY USE ONLY (Leave Blank)	2. REPORT DATE	3. REPORT TYPE AND DATES COVERED	
4. TITLE AND SUBTITLE		5. FUNDING NUMBERS	
Communication, Feedback and Decentralized Control for Underwater Vehicules		N0001400WR20277	
6. AUTHOR(S)			
Daniel J. Stilwell Bradley D. Bishop			
7. PERFORMING ORGANIZATION NAME(S) AND ADDRESS(ES)		8. PERFORMING ORGANIZATION REPORT NUMBER	
Division of Engineering and Weapons United States Naval Academy Annapolis, MD 21402-5042		ES-05-00	
9. SPONSORING/MONITORING AGENCY NAME(S) AND ADDRESS(ES)		10. SPONSORING/MONITORING AGENCY REPORT NUMBER	
Office of Naval Research Ballston Tower One 800 North Quincy Street Arlington, VA 22217-5660			
11. SUPPLEMENTARY NOTES			
12a. DISTRIBUTION/AVAILABILITY STATEMENT		12b. DISTRIBUTION CODE	
Approved for public release; distribution unlimited			
13. ABSTRACT (Maximum 200 words)			
<p>Since communications bandwidth underwater is severely limited, platoons of underwater vehicles must cooperate with a minimum of direct communication. We present a control strategy for a platoon of underwater vehicles that requires broadcast-only communications from a single vehicle within the platoon. Even as the size of the platoon increases, the bandwidth of required communication remains constant. Our emphasis is on the mathematical developments necessary to prove existence of such a control system. In addition, we briefly describe a control system design procedure and present a simulation-based example.</p>			
14. SUBJECT TERMS		15. NUMBER OF PAGES	
Underwater vehicles, control systems, decentralized control		31	
		16. PRICE CODE	
17. SECURITY CLASSIFICATION OF REPORT	18. SECURITY CLASSIFICATION OF THIS PAGE	19. SECURITY CLASSIFICATION OF ABSTRACT	20. LIMITATION OF ABSTRACT
Unclassified	Unclassified	Unclassified	UL

NSN 7540-01-280-5500

Standard Form 298, (Rev. 2-89)
Prescribed by ANSI Std. Z39-18
299-01